

ANSWERS SAMPLE EXAM AE2103 SPACE MISSIONS AND APPLICATIONS I

Question 1	True, geostationary is a special case of geosynchronous.
Question 2	<p>200 orbits in 8 days = 15 orbits/day. $25 \cdot 2\pi = 50\pi$ rad/day.</p> $\frac{50\pi}{24 \cdot 60 \cdot 60} = 0,00182 \text{ rad/s}$ <p>Therefore the angular velocity has to be higher to achieve the repeat orbit and the requirement is not met.</p>
Question 3	True
Question 4	True
Question 5	False, a geoid is not a mathematically perfect ellipsoid
Question 6	<p>First we derive A by assuming $\Delta S_{L1} = \Delta S_{L2}$</p> <p>Hence we obtain</p> $\Delta t_{L1} - \frac{A}{f_{L1}^2} = \Delta t_{L2} - \frac{A}{f_{L2}^2}$ <p>Rearrange terms.</p> $\frac{A}{f_{L1}^2} - \frac{A}{f_{L2}^2} = \Delta t_{L1} - \Delta t_{L2}$ <p>Put the A in just one fraction</p> $\frac{Af_{L2}^2 - Af_{L1}^2}{f_{L1}^2 f_{L2}^2} = \Delta t_{L1} - \Delta t_{L2}$ $A = \frac{f_{L1}^2 f_{L2}^2 (\Delta t_{L1} - \Delta t_{L2})}{f_{L2}^2 - f_{L1}^2}$ <p>Now put A into the equation for Δs</p> $\Delta s = c_{vac} \left[\Delta t_{L1} - \frac{A}{f_{L1}^2} \right] = c_{vac} \left[\Delta t_{L1} - \frac{f_{L1}^2 f_{L2}^2 (\Delta t_{L1} - \Delta t_{L2})}{f_{L1}^2 (f_{L2}^2 - f_{L1}^2)} \right]$ $\Delta s = c_{vac} \left[\Delta t_{L1} - \frac{f_{L2}^2 (\Delta t_{L1} - \Delta t_{L2})}{f_{L2}^2 - f_{L1}^2} \right]$
Question 7	<p>The maximum errors occurs when 1 accelerometer has an error of $1,5e^{-12}$ and the other has an error of $-1,5e^{-12}$. Then we obtain:</p> $E_{total} = 3e^{-12}/0,5 = 6e^{-12} \text{ m/s}^2 = 6e^{-3} \text{ E} = 6 \text{ mE}$
Question 8	$e = \left(\frac{r_{sat}}{R_e} \right)^2 \cdot 1,5E^{-12} = 1,62E^{-12} \text{ m/s}^2$

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Question 9	The satellite moves 2π in one year. Hence $\frac{2\pi}{265,25 \cdot 24 \cdot 60 \cdot 60} = 1,991 \cdot 10^{-7} \frac{rad}{s}$
Question 10	<p>First we calculate the velocity</p> $V = \sqrt{\frac{\mu}{r}} = \sqrt{\frac{3986 \cdot 10^2}{(6378 + 1300)}} = 7,2 \text{ km/s}$ <p>1 orbit takes $2\pi(6378+1300)/7,2 = 6700$ s to complete In 10 days the satellite makes 129 orbits $2\pi(6378)/129=311$ km per orbit</p> <p>As the earth rotates during these 10 days we have to take that into account. $360/(365,25 \cdot 24 \cdot 60 \cdot 60) \cdot 6700 = 0,076$ deg/orbit = 1,333 mrad/orbit $1,333 \text{ m} \cdot 6378 = 8,5$ km Adding these up gives a width of 319 km. The difference between this answer and the answers provided is caused because of the use of sidereal days vs regular das. Answer a is correct</p>
Question 11	$\lambda = \frac{c}{f} = \frac{3 \cdot 10^8}{1,25 \cdot 10^9} = 0.24 \text{ cm}$ $q = \frac{\lambda}{L} = \frac{0.24}{10} = 0.024 \text{ rad}$ $0.024 \cdot \frac{180}{\pi} = 1,37 \text{ deg}$
Question 12	Infrared had the highest wavelength thus the lowest frequency. Yellow is between red and blue. Hence all answers cancels except for c
Question 13	All of the above.